DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

A38CE Revision 9 BEECH 2000 February 15, 1999

TYPE CERTIFICATE DATA SHEET NO. A38CE

This data sheet which is part of the Type Certificate No. A38CE prescribes conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder: Raytheon Aircraft Company Wichita, Kansas 67201

I. Model 2000, Starship 1, (Commuter Category), approved June 14, 1988

Engine Two (2) United Aircraft of Canada, Ltd., or Pratt & Whitney of Canada PT6A-67A

(turboprop) per P&WC Specification No. 950.

Fuel JET A, JET A-1, and JET B conforming to P&WC S.B. 132044(2)(3) or ASTM

Spec. D-1655; JP-4, JP-5 (MIL-T-5624); JP-8 (MIL-T-83133). See FAA Approved

Airplane Flight Manual (AFM) for emergency fuels.

Fuel not containing icing inhibitors must have MIL-I-27686 fuel system icing inhibitor added in amounts of not less than 0.06%, or more than 0.15% by volume. The FAA Approved Airplane Flight Manual (AFM) contains procedures for

blending icing inhibitor to fuel.

Oil (Engine & Gearbox) UACL PT6 Engine Service Bulletin No. 14001 lists approved brand oils.

Engine limits

			N1 Gas	Prop	Max. Permissible
	Shaft		Generator	Shaft	Turbine Interstage
	HP	Torque	Speed	Speed	Temp. (Deg. C)
Takeoff (5 min.)	1200	*100%	104%	1700	850°
Max. cont. (1)	1200	*100%	104%	1700	840°
Max. cont. (2)	1193	*100%	104%	1690	840°
Starting transient					
(5 sec.)					1000°
*100% torque - 37	07 ft -1h				

^{*100%} torque - 3707 ft.-lb.

At low altitude and low ambient temperature the engines may produce more power at takeoff than that for which the airplane has been certified. Under these conditions the displayed torque limitations shall not be exceeded. The AFM provides minimum torque settings for T.O. It must be possible to achieve these settings without exceeding ITT or N_1 limits.

Oil temperature: -40° C minimum starting

-40° C to 110°C idle

 0° C to 110° C max. continuous

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⁽¹⁾ NC-4 thru NC-20

⁽²⁾ NC-21 and on and NC-4 thru NC-20 with Beech Kit 122-3003

I. Model 2000 (cont'd)

Propeller and propeller limits

Two (2) McCauley 5JFR36C1003 hubs, each with five (5) McCauley D-L104DSZ-0

blades

Diameter: 104 in. (maximum, no reduction permitted

Pitch settings at:

Flight idle stop
Ground idle stop
Feathered

LH
RH
-----See Note 4---------See Note 5-----90.9°
91.7°

Continuous ground idle between 700 and 1000 rpm is prohibited.

Continuous flight operation between 1450 and 1580 rpm is prohibited.

Airspeed limits (IAS)

S/N NC-4 thru NC-28

Max. operating speed, V_{MO} 250 knots up to 10,000 ft.

Linear variation between 10,000 ft. and

12,000 ft. to 270 knots

270 knots from 12,000 ft. to 21,000 ft.

Max. operating Mach No., M_{MO} .60 (above 21,000 ft.)

S/N NC-29 and on and NC-4 thru NC-28 with Beech Kit 122-9002

Max. operating speed, V_{MO} 245 knots up to 10,000 ft.

Linear variation between 10,000 ft. and

12,000 ft. to 265 knots

265 knots from 12,000 ft. to 21,900 ft.

Max. operating Mach No., M_{MO} .60 (above 21,900 ft.)

S/N NC-4 and on

Maneuvering speed, V_A 175 knots (S/N NC-4)

Maneuvering speed, V_A 181 knots (S/N NC-5 and on and NC-4 with

Beech Service Bulletin No. 2343)

Gust penetration speed, V_B 192 knots

Max. flap extension speed

 $\begin{array}{ccc} (100\% \ position, \ 14^\circ), \ V_F & 180 \ knots \\ Landing \ gear \ extended, \ V_{LE} & 200 \ knots \\ \end{array}$

Landing gear operating, V_{LO} 200 knots extending Landing gear operating, V_{LO} 180 knots retracting

C.G. Range (landing gear extended)

S/N NC-4

FS 314.6 to FS 320.0 at 14,400 lb.

FS 308.0 to FS 320.0 from 13,575 lb. and below

S/N NC-5 thru NC-20 and NC-4 with Beech Service Bulletin No. 2343

FS 309.4 to FS 320.0 at 14,400 lb.

FS 307.0 to FS 320.0 from 13,450 lb. and below

S/N NC-21 thru NC-28 and NC-4 thru NC-20 with Beech Kit 122-3003

FS 309.7 to FS 320.0 at 14,500 lb.

FS 307.0 to FS 320.0 from 13,450 lb. and below

S/N NC-29 and on and NC-4 thru NC-28 with Beech Kit 122-9002

FS 310.9 to FS 320.0 at 14,900 lb.

FS 307.0 to FS 320.0 from 13,450 lb. and below

Straight line variation between given points

Moment change due to flap/forward wing sweep and retraction of landing

gear = +2316 in-lb.

Empty Weight C. G. Range

I. Model 2000 (cont'd) Datum	86.2 inches forward of the center of the front jack point located at aft end of the nose gear door opening.				
Leveling means	The aircraft is leveled with a plumb bob using leveling points located at FS 176.40. The upper point is in the top main cabin door frame just forward of the top latch pin hole. The lower leveling point is in the bottom cabin door frame.				
Maximum Weight	Ramp Takeoff Landing Zero fuel (1) NC-4 thru N (2) NC-21 thro (3) NC-29 and	ugh NC-28 and			
Minimum Crew	One pilot; or One pilot and one copilot See Note 6				
No. of Seats & Cargo Landing	Maximum 11 including crew at FS 143. See loading instructions in AFM for approved seating and cargo configurations.				
Maximum baggage	160 lbs. at FS 182 (forward) 250 lbs. at FS 340 (folded aft couch) 525 lbs. at FS 386 (aft)				
Fuel capacity	Tank	Cap. (ga	ıl.) Usab	ole (gal.)	Arm
	S/N NC-4 thru N Left side Right side	VC-28 269.0 269.0	267.0 267.0		+322.2 +322.2
	S/N NC-29 and a Left side Right side	on and NC-4 th 284.5 284.5	nru NC-28 with 282.: 282.:	5	22-9002 +321.0 +321.0
Oil capacity	32 qt. total (29 q engine tanks)	uart drainable) at FS 447.4 (i	includes 12 qt	. usable in two integral
Maximum operating altitude	41,000 ft.				

<u>I. Model 2000</u> (cont'd) Control surface movements

	<u>Trailing Edge Position</u>			
Surface	Up/Outboard	<u>Neutral</u>	Down/Inboard	
Elevon tabs	10°	2° TEU	6°	
Right elevon				
Roll deflections				
Pitch full aft	33°	9° TEU	9°	
Pitch neutral	19°	2° TED	16°	
Pitch full fwd	15°	5° TED	17°	
Left elevon				
Roll deflections				
Pitch full aft	37°	13° TEU	5°	
Pitch neutral	23°	2° TEU	12°	
Pitch full fwd	19°	1° TED	13°	
Elevator tabs***(1)	18°	0°	2°	
Elevator tabs***(2)	20.5°	2.5° TEU	-0.5°	
Elevator**	13°	0°	28°	
Rudder tabs	18°	0°	18°	
Rudder	34°	0°	34°	
Wing flap maximum	14°			
Forward wing position				
(Sweep angle related	Flaps down	-4° (fwd sw	/eep)	
to flap position)	Flaps up	30° (aft swe	eep)	

- * Nominal Values See Model 2000 Maintenance Manual for rigging instructions, deflections, and corresponding tolerances.
- ** Elevator TEU = A/C pitch down; TED = A/C pitch up
- *** Initial rigging only. Final rigging is by flight test in accordance with the Model 2000 Maintenance Manual.
- (1) NC-4 thru NC-28
- (2) NC-29 and on and NC-4 thru NC-28 with Beech Kit 122-9002.

Serial numbers eligible

NC-4 and after

Certification basis

FAR Part 23 effective February 1, 1965, as amended by Amendments 23-1 through 23-34; Special Federal Aviation Regulation (FAR) No. 27, effective February 1, 1974, as amended by Amendments 27-1 through 27-5; FAR Part 36 as amended by Amendments 36-1 through 36-15; Special Conditions No. 23-ACE-11 and Exemption No. 4827 from compliance with Section 23.807(d)(1)(i), Exemption No. 5094 from compliance with Section 23.207(c) and Exemption 5411 from Sections 23.473(c) and 23.1001. Compliance with ice protection has been demonstrated in accordance with FAR 23.1419.

Equivalent Safety Items

(1) Landing Gear WarningFAR 23.729(f)(2)(2) RPM Control KnobFAR 23.781(b)(3) Fuel Pressure GaugeFAR 23.1305(g)

(4) Powerplant Instrument

Markings Torque/ITT FAR 23.1549(b) and (c)

Application for Type Certificate dated February 2, 1987. Type Certificate No. A38CE issued June 14, 1988, obtained by manufacturer under Delegation Option Procedures.

Production Basis

S/N NC-5, NC-11 and after, Production Certificate No. 8

S/N NC-4 and NC-6 through NC-10, prior to original certification of each aircraft, an FAA representative performed a detailed inspection for workmanship, materials, and conformity with the approved technical data, and a check of the flight characteristics.

Delegation Option Manufacturer No. CE-2 authorized to issue airworthiness certificate under delegation provisions of Part 21 of the Federal Aviation Regulations.

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The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification.

In addition, the following items of equipment are required.

 Crew member oxygen mask, Beech P/N 101-384220-5, or EROS, P/N MC-10-15-10, or Beech P/N 101-384220-7, or SCOTT-EROS P/N MC-10-15-12, or Beech P/N 101-384220-9, or SCOTT-EROS P/N MC-10-15-18.

- NOTE 1. Current weight and balance data, loading information, and a list of equipment included in empty weight must be provided for each airplane at the time of original certification.
 - (a) Basic empty weight includes unusable fuel of 38.0 lbs. at (326.0 in.) with 10.5 lbs. being undrainable.
 - (b) Basic empty weight includes engine oil of 58.5 lbs. at (447.4 in.).
- NOTE 2. All placards required in the FAA Approved Airplane Flight Manual (AFM) must be installed in the appropriate location. The following AFM's are applicable:

Beech Part Number 122-590013-3G (S/N NC-4 thru NC-20 only)

Beech Part Number 122-590013-23 (S/N NC-21 thru NC-28; NC-4 thru NC-20 with Beech Kit

122-3003)

Beech Part Number 122-590013-37 (S/N NC-29 and on; NC-4 thru NC-28 with Beech Kit 122-9002)

- NOTE 3. Mandatory retirement times and inspection times are contained in the FAA Approved Limitations Section, Chapter 4, of the Model 2000 Maintenance Manual. These limitations may not be changed without FAA approval.
- NOTE 4. The flight idle stop is set to absorb 17.5% torque at 1500 propeller rpm, sea level std. day.
- NOTE 5. The ground idle stop is set to absorb 9.5% torque at 1500 propeller rpm, sea level std. day.
- NOTE 6. Airplane S/N's NC-4 through NC-22 must be modified by Beechcraft Kit P/N 122-3001 prior to operations with one pilot. Except where otherwise prescribed by the appropriate operating regulations:
 - (a) One pilot with a BE2000S type rating. Pilot must adhere to a single pilot equipment requirements contained in the Airplane Flight Manual Kinds of Operations Equipment List; or
 - (b) One pilot and one co-pilot. Pilot must have a BE-2000 or BE-2000S type rating.

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